

RD-A166 828

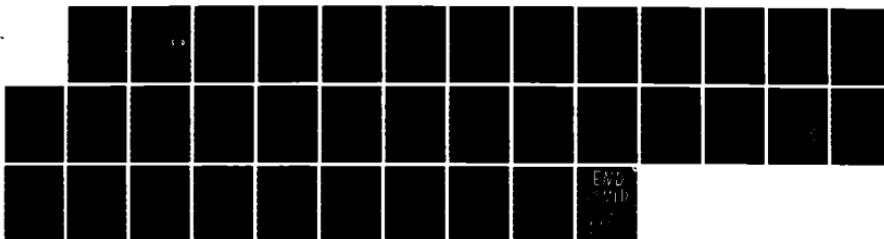
PULL-OFF FORCES FOR ADHESIVE TAPES(U) AKRON UNIV OH  
INST OF POLYMER SCIENCE A N GENT ET AL. APR 86 TR-5  
N00014-85-K-8222

1/1

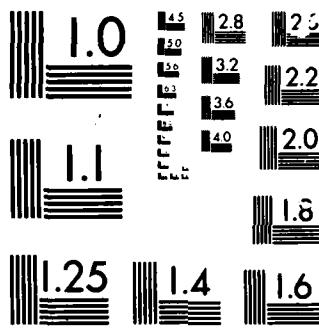
UNCLASSIFIED

F/G 11/1

ML



END  
10/10/86



MICROCOM<sup>®</sup>

CHART

AD-A166 820

(12)

OFFICE OF NAVAL RESEARCH  
Contract N00014-85-K-0222  
Project NR 092-555

Technical Report No. 5

PULL-OFF FORCES FOR ADHESIVE TAPES

by

A. N. Gent and S. Kaang

DTIC  
ELECTE  
APR 17 1986  
S D

Institute of Polymer Science  
The University of Akron  
Akron, Ohio 44325

April, 1986

Reproduction in whole or in part is permitted for  
any purpose of the United States Government

Approved for Public Release; Distribution Unrestricted

DTIC FILE COPY

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER Technical Report No. 5	2. GOVT ACCESSION NO. ADN 166800	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) Pull-Off Forces for Adhesive Tapes	5. TYPE OF REPORT & PERIOD COVERED Technical Report	
7. AUTHOR(s) A. N. Gent and S. Kaang	6. PERFORMING ORG. REPORT NUMBER N00014-85-K-0222	
9. PERFORMING ORGANIZATION NAME AND ADDRESS Institute of Polymer Science The University of Akron Akron, Ohio 44325	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS NR 092-555	
11. CONTROLLING OFFICE NAME AND ADDRESS Office of Naval Research Power Program Arlington, VA 22217	12. REPORT DATE April, 1986	
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)	13. NUMBER OF PAGES 33	
	15. SECURITY CLASS. (of this report) Unclassified	
	16a. DECLASSIFICATION/DOWNGRADING SCHEDULE	
16. DISTRIBUTION STATEMENT (of this Report) According to attached distribution list. Approved for public release; distribution unrestricted.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES Submitted for publication in: Journal of Applied Polymer Science		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Adhesion, Adhesives, Detachment, Fracture energy, Fracture mechanics, Peeling, Separation, Strength.		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) An analysis is given of the force $F$ required to pull an adhesive tape of unit width away from a rigid substrate in terms of the strength $G_a$ of adhesion, the tensile modulus $E$ of the tape, and its thickness $t$ . Measurements are reported for several commercial adhesive tapes and compared with the predictions of the theory. Excellent agreement is		

obtained, suggesting that the theory is basically correct. Attention is drawn to the unusual form of the dependence of the failure force  $F$  upon the work  $G_a$  of detachment and the resistance  $E_t$  of the tape to stretching in this case:  $F \propto E_t G_a^3$ . Even though the tape is assumed to be linearly-elastic, the markedly non-linear (cubic) relation between force  $F$  and displacement  $\delta$  of the tape away from the substrate leads to this unusual result. Differences observed in  $G_a$  from pull-off and from 90° peeling experiments are tentatively attributed to additional energy losses in the latter case due to the severe bending deformations imposed on the tape as it is peeled away.

d. a

variation

delta

? Keywords:

FLD 19

### 1. Introduction

When adhesive tapes are pulled away from a rigid substrate, as shown schematically in Figure 1, the force required depends upon both the strength of adhesion and the resistance of the tape to stretching. Although these two factors are obviously significant, no previous analysis of their relative importance is known to the present authors. A simple theoretical treatment is therefore given below relating the pull-off force  $F$  to the strength of adhesion, characterized by the work  $G_a$  required to detach unit area of adhering tape from the substrate, and the effective tensile (Young's) modulus  $E$  of the tape, assumed for simplicity to be linearly elastic. Measurements with various commercial tapes are then reported, and compared with the theoretical predictions.

Because of the simplicity of this experiment, and the ready way in which values of  $G_a$  and  $E$  can be deduced from it, it may have potential value as a routine test method for adhesive tapes. This is particularly the case for tapes that are commonly used to secure items to a rigid base, when the pull-off force  $F$  represents an important service parameter.

Quite apart from any potential practical value, the analysis of the pull-off force  $F$  has some scientific interest, for two reasons. It demonstrates once again the power of simple energy considerations in fracture mechanics, using a characteristic value of the detachment energy  $G_a$  as the criterion for debonding (1-6). And the pull-off force  $F$  is found to be neither proportional to  $G_a$ , as might at first be expected and is, indeed, observed in simple peeling experiments (7-9), nor is it proportional to  $(EG_a)^{\frac{1}{2}}$  as is found in many linearly-elastic ("Griffith") systems where energy is expended in deforming layers after debonding them (10-13). Instead, it is found to be proportional to  $(EG_a^3)^{\frac{1}{2}}$ , a result

## 3.

which emerges directly from the analysis as a consequence of the particular relation which holds between the force  $F$  and the corresponding elastic displacement  $\delta$  of the tape when no further debonding occurs;  $F \propto \delta^3$ ; even though the components are assumed to be linearly-elastic (14). This is the first time to the authors' knowledge that other possible types of dependence of the failure force  $F$  upon  $E$  and  $G_a$  have been pointed out.

2. Theoretical considerations(i) Elastic behavior

A sketch of an adhesive tape being pulled away from a rigid substrate is shown in Figure 1. The tensile strain  $e$  in the tape is obtained in terms of the angle  $\theta$  between the detached part of the tape and the substrate surface from geometrical considerations:

$$e = \sec \theta - 1. \quad (1)$$

Thus, when  $\theta$  is small,

$$e \approx \theta^2/2. \quad (2)$$

The tensile force  $F'$  in the detached part of the tape is related to the applied pull-off force  $F$ ,

$$F' = 2F \sin \theta. \quad (3)$$

Assuming that the tape is linearly-elastic, with an effective value of tensile (Young's) modulus  $E$ , the force  $F'$  is given by  $Eewt$ , where  $w$  and  $t$  are the width and thickness of the tape.

Thus,

$$F' = 2Ewt \sin \theta (\sec \theta - 1).$$

When  $\theta$  is small, this simplifies to yield (14):

$$F = Ewt \theta^3.$$

(4)	
(5)	
Distribution	
Availability Codes	
Dist	Avail. a.d/or Special
A-1	

QUALITY  
CONTROLLED  
3

## (ii) Conditions for detachment

We now consider the energy changes that take place for further detachment by a distance  $2\Delta c$  (Figure 1). Work supplied by the pull-off force  $F$  is  $F \tan \theta \Delta c$ . Work expended in detachment is  $2G_a w \Delta c$ , and work expended in stretching the newly-detached parts of the tape is  $Ee^2 w t \Delta c = Ewt(\sec \theta - 1)^2 \Delta c$ . By equating the work supplied to the total work expended we obtain

$$F \tan \theta = 2G_a w + Ewt(\sec \theta - 1)^2. \quad (6)$$

On substituting for  $F$  from equation (4) and rearranging:

$$G_a/Et = \frac{1}{2} \tan^2 \theta + \cos \theta - 1. \quad (7)$$

When  $\theta$  is small this becomes

$$G_a/Et = 3\theta^4/8. \quad (8)$$

Equations 7 and 8 give the work  $G_a$  of detachment in terms of the angle  $\theta$  between the detached tape and the substrate. In terms of the pull-off force  $F$  and angle  $\theta$  from equation 5,

$$G_a = (3/8) F \theta/w, \quad (9)$$

and in terms of  $F$  and the tape modulus  $E$ ,

$$F/w = (8G_a/3)^{3/4} (Et)^{1/4}. \quad (10)$$

These results are valid only at small values of  $\theta$ , because they depend upon the approximations leading to equations 5 and 8. The exact result for  $F$  is given in parametric form by equations 4 and 7. However, even for values of the angle  $\theta$  as large as  $45^\circ$  the error is less than 10 per cent when  $G_a$  is calculated from equation 10 because of compensating errors in equations 5 and 8. On the other hand, if  $G_a$  is calculated from measurements of  $\theta$  by means of equation 8 or 9, then the error is about

5.

10 per cent when  $\theta$  is  $25^\circ$  and becomes rapidly greater for larger angles.

In the following parts of the paper experimental measurements of the pull-off force  $F$  and angle  $\theta$  are described for some pressure-sensitive adhesive tapes and compared with the theoretical relations given above.

3. Experimental details(i) Materials

Several commercial pressure-sensitive adhesive tapes were employed in the experiments:

- A, a vinyl plastic electrical tape, 19 mm wide and about 0.235 mm thick (3M Company, denoted 88)
- B, a window film mounting tape, 12.7 mm wide and about 0.105 mm thick (3M Company, Catalog No. 2145)
- C, a relatively thick, soft, extensible mounting tape, 12.7 mm wide and about 1.34 mm thick (3M Company, Catalog No. 110)
- D, a clear tape, 25.4 mm wide and about 0.14 mm thick (Manco Tape Inc., denoted All-Weather Clear Tape)
- E, a paper-based masking tape, 25.4 mm wide and about 0.145 mm thick (Tuck Tape)

(ii) Tensile stress-strain relations

Measurements were made of the relations between tensile force per unit width and extension for the first three tapes, using strips about 300 mm long, stretched at 5 mm/min. They were approximately linear for tapes A and C over the range 0-20 per cent extension, Figure 2a, but highly non-linear for tape B, which underwent plastic yielding at about 3 per cent extension, Figure 2b. Values of the average tensile strains set up during pull-off experiments<sub>A</sub> were deduced from the measured pull-off angles  $\theta$  by means of equation 2; they were 5.0 per cent for tape A, 2.3 per cent for tape B and 13.3 per cent for tape C. Effective values of Et were calculated from the corresponding tensile stresses of 3.50 kN/m, 85.5 kN/m and 1.25 kN/m, respectively. (Using the measured tape thicknesses t, these results correspond from glass

to effective values of tensile modulus  $E$  of 15 MPa, 820 MPa and 0.92 MPa for tapes A, B and C.)

Because the detachment forces with a Teflon substrate were significantly smaller for Tapes B and C, the average tensile strains were also smaller, about 0.8 per cent and about 3.8 per cent, respectively, and the effective values of  $E_t$  were correspondingly somewhat larger than before, about 105 kN/m and about 1.5 kN/m, due to the non-linear stress-strain relations.

(iii) Measurement of pull-off forces

Samples of tape about 350 mm long were applied to a rigid horizontal substrate, a polished glass plate or a smooth Teflon plate, previously cleaned with acetone. A stiff wire loop, trapped between the center of the strip of tape and the substrate, was then used to pull the tape away. Pull-off forces  $F$  and angles  $\theta$  were measured as shown schematically in Figure 1, with a tensile testing machine. To prevent the tape from slipping along the substrate during pull-off, the ends were wrapped around the ends of the substrate plate and in some instances secured there by tape clamps. In order to vary the effective stiffness  $E_t$  without changing the detachment energy  $G_a$ , up to ten layers of tape were applied, one on top of another. On the other hand, by using the same tapes on two different substrates, glass and Teflon, it was hoped to vary  $G_a$  substantially without changing the effective stiffness of the tape.

As the tape began to pull away from the substrate the applied force  $F$  rose to a relatively-large starting value and then fell to a value about

30 per cent lower and remained at this level as detachment continued over long distances. Steady-state values of F and the pull-off angle  $\theta$  have been taken here as representative of pull-off at a constant rate of detachment. The initial surge is ascribed to higher start-up velocities.

All experiments were carried out at ambient temperature, about 24°C, and with a crosshead speed of 83  $\mu\text{m/s}$ .

(iv) Independent measurements of  $G_a$

Measurements were made of the force F required to peel tapes away from the substrates at an angle of 90°, Figure 3, and at various speeds v in the range 0.1 to 1 mm/s. Values of detachment energy  $G_a$  were then calculated:

$$G_a = F/w. \quad (11)$$

By interpolation, values were obtained appropriate to the speed  $dc/dt$  at which debonding took place in the pull-off experiments, where  $dc/dt = v / \tan \theta$ , Figure 1.

4. Experimental results and discussion(i) Pull-off forces and angles

Measured values of pull-off force  $F$  and angle  $\theta$  are given in Tables 1 and 2. Values of detachment energy  $G_a$  calculated from them by means of equation 9 are given in the fourth column of Tables 1 and 2 and values calculated from the pull-off force  $F$  alone, with the separately-determined value of the effective tensile stiffness  $E_t$  for each tape 1, using equation 10, are given in the fifth column of Tables 1 and 2. These two estimates of  $G_a$  are in reasonable agreement with each other in all cases, suggesting that the essential features of the mechanics of pulling away an extensible tape from a rigid substrate are contained in the theoretical treatment. However, they are not generally in good agreement with direct measurements of  $G_a$  by peeling away the tape at an angle of  $90^\circ$ , given in the final columns of Tables 1 and 2 for peel velocities equal to the computed rates of advance of the separation front in the pull-off experiments. The discrepancies are significant, and rather different in magnitude for the different tapes. For tapes A and D, for example, the peel energies are about 2X to 3X the pull-off energy, whereas for tapes C and E and for tape B adhering to Teflon, the peel energy is closer to the pull-off energy. Possible reasons for these differences are discussed later. We note here only that values of detachment energy  $G_a$  obtained from pull-off experiments are internally consistent and generally lower than those obtained from peeling experiments.

A striking feature of the present theoretical treatment is the form of the predicted dependence of pull-off force  $F$  upon the effective thickness of the adhering tape  $t$ ;  $F \propto t^{\frac{1}{2}}$ ; equation 10. Experimental values of  $F$  are plotted in Figures 4 and 5 against  $N^{\frac{1}{2}}$ , where  $N$  is the number of layers of tape applied one on top of another and pulled away together. Clearly, the effective tape thickness  $t$  is proportional to  $N$  in these experiments. As can be seen in Figures 4 and 5, accurately linear relations were obtained between  $F$  and  $N^{\frac{1}{2}}$  in all cases, in good accord with the theoretical prediction.

A further prediction of the theory is that the product  $F\theta$  will be independent of the stiffness of the tape, and hence of the thickness  $t$  or number  $N$  of layers pulled off together (except insofar as the speed of separation is altered, so that changes are brought about in the detachment energy  $G_a$  on this account). Values of  $F\theta$  are plotted in Figures 6, 7 and 8 against the number  $N$  of adhering layers. They are seen to be substantially constant, independent of  $N$ , even though  $F$  and  $\theta$  vary separately with  $N$  to a significant extent, Tables 1 and 2.

It is interesting to note that the apparent detachment energy  $G_a$ , given by  $3F\theta/8w$ , was approximately the same for Tape A pulled away from a glass or a Teflon surface. In contrast, for Tapes B and C the detachment energies for a Teflon surface were only about 25 per cent and 15 per cent of those for a glass surface, in accord with the lower wettability expected for Teflon. The adhesion of Tape A must be attributed largely to its rheological features rather than to selective wettability.

(ii) Discrepancies in  $G_a$ 

Several possible reasons may be adduced for the observed discrepancies in the detachment energy  $G_a$  from pull-off and from peeling experiments. In the first place, equations 9 and 10 are based on the assumption that the pull-off angle  $\theta$  is small. This is not always a valid assumption, especially for strongly-adhering, easily-stretched tapes, Tables 1 and 2. However, the values obtained from equations 9 and 10 are in good agreement, even though the assumption of small  $\theta$  is more stringent in the first case. Also, the discrepancy is not markedly reduced when many layers of tape are detached together and the angle  $\theta$  is much smaller. Finally, the size of the discrepancy does not correlate well with the magnitude of  $\theta$ . We conclude that the simplifying assumption that  $\theta$  is small is not responsible for the observed discrepancies.

A second possible cause is non-linear elastic behavior of the tapes in tension. In contrast to the assumed linear elastic response, the tapes followed a non-linear relation between tensile force and elongation to various degrees, Figure 2, so that the effective stiffness  $E_t$  at small strains and pull-off angles was greater than at large ones. It seems probable that the use of an average value of  $E_t$  in calculating  $G_a$  from pull-off experiments is responsible for a small but systematic change in the values obtained as the number of layers was increased and the imposed tensile strain was correspondingly reduced. This feature should be most pronounced for tapes which yield in tension, Tapes B and D, and at large values of  $\theta$ , i.e; for pull-off of single layers.

But these results do not seem to be particularly anomalous, Tables 1 and 2. It must therefore be concluded that the simplifying assumption of linearly-elastic behavior, although quite inadequate for tapes which undergo plastic yielding, was a reasonably satisfactory approximation in most of the experiments reported here.

A third assumption implicit in the theoretical treatment is that work expended in bending the tape away from the substrate is negligible, or at least is the same in both the pull-off and the peeling experiments so that it contributes equally to the values obtained for  $G_a$ . In some circumstances this contribution can be both large and strongly dependent upon the magnitude of the peel angles (15). It would also be expected to depend upon the structure of the tape and hence to vary from one tape to another. Thus, it may be the primary factor responsible for the observed discrepancies in  $G_a$  from pull-off at small angles and from peeling at  $90^\circ$ , even though the mode of failure appears to be so similar in the two cases. Further work is needed to clarify this point.

### 5. Conclusions

The predicted dependence of the pull-off force upon the effective stiffness  $E_t$  of the tape, the number  $N$  of layers applied, and the type of substrate used were found to hold reasonably well. In particular, the unusual forms of the predicted dependence upon  $(E_t)^{\frac{1}{2}}$  and upon  $G_a^{\frac{3}{4}}$  appear to be correct. Thus, the pull-off experiment appears to be a simple way of characterizing both the energy  $G_a$  required for detaching an adhesive tape at small angles and the effective tensile stiffness of the tape. Moreover, it resembles many service applications of pressure-sensitive tapes. If a tape stretches too much, so that the angle  $\theta$  becomes unreasonably large (greater than about  $30^\circ$ , say) then two or more layers of tape can be applied and pulled off together. In some instances it was found that the layers did not adhere to each other as well as they adhered to the substrate; the multi-layer method is then not a feasible way of reducing  $\theta$  to sufficiently small values and the parametric solutions for  $F$  must be employed.

### Acknowledgements

This work forms part of a program of adhesion research at The University of Akron supported by the Adhesive and Sealants Council. Additional support by the Office of Naval Research (Contract N00014-85-K-0222) is gratefully acknowledged. The authors are also indebted to Professor Roger Fosdick, Department of Aerospace Engineering and Mechanics, University of Minnesota, for helpful discussions on non-linear elastic behavior.

References

1. Rivlin, R.S., Paint Technol. 9, 215 (1944).
2. Ripling, E.J., Mostovoy, S., and Patrick, R.L., Materials Res. Stand. 4, 129 (1963).
3. Malyshev, B.M., and Salganik, R.L., Int. J. Fracture Mech. 1, 114 (1965).
4. Williams, M.L., Proc. 5th U.S. Natl. Congress on Applied Mechanics, Minneapolis, June, 1966, ASME, New York, 1966, p.451.
5. Gent, A.N., and Kinloch, A.J., J. Polym. Sci., Part A-2 9, 659 (1971).
6. Kendall, K., Proc. Roy. Soc. Lond. A344, 287 (1975).
7. Ahagon, A., and Gent, A.N., J. Polym. Sci: Polym. Phys. Ed. 13, 1285 (1975).
8. Kendall, K., J. Phys. D: Appl. Phys. 8, 512 (1975).
9. Chang, R.J., and Gent, A.N., J. Polym. Sci: Polym. Phys. Ed. 19, 1619 (1981).
10. Williams, M.L., J. Appl. Polym. Sci. 13, 29 (1969).
11. Kendall, K., J. Materials Sci. 11, 638 (1976).
12. Anderson, G.P., Bennett, S.J., and DeVries, K.L., "Analysis and Testing of Adhesive Bonds", Academic Press, New York, 1977.
13. Gent, A.N., Fielding - Russell, G.S., Livingston, D.I., and Nicholson, D.W., J. Materials Sci. 16, 949 (1981).
14. Southwell, R.V., "An Introduction to the Theory of Elasticity", 2nd. ed., Oxford Univ. Press, 1941, p. 19.
15. Gent, A.N., and Hamed, G.R., J. Appl. Polym. Sci., 21, 2817 (1977).

Table 1: Detachment from glass

Number of layers <u>N</u>	Pull-off force F/w (N/m)	Pull-off angle $\theta$ (rad)	From eq.9	$G_a$ (N/m)		
				From eq.10	From eq.11	
Tape A						
1	217±5	0.42	34	32	95	
2	268±10	0.34	34	33.5	100	
3	320±10	0.31	37	37	102	
4	360±20	0.30	40.5	39.5	104	
5	445±20	0.29	48.5	48.5	105	
6	455±20	0.27	46	47	107	
7	475±20	0.26	46.5	47.5	108	
8	545±20	0.255	52	54.5	109	
9	550±20	0.245	50.5	53	110	
10	558±20	0.235	49	52	112	
Tape B						
1	1585±15	0.36	214	157	290	
2	2110±20	0.255	202	183	305	
3	2400±155	0.225	202	190	315	
4	2665±155	0.20	201	198	320	
5	2550±230	0.175	167	174	330	
6	2705±230	0.165	167	176	335	
7	3170±310	0.165	197	208	335	
8	3245±310	0.165	201	205	335	
9	3630±310	0.165	225	228	335	
10	3475±310	0.165	216	208	335	

Table 1 (continued)

Number of layers <u>N</u>	Pull-off force <u>F/w</u> (N/m)	Pull-off angle <u>θ</u> (rad)	From eq.9	<u>G<sub>a</sub></u> (N/m)	
				From eq.10	From eq.11
Tape C					
1	340±25	0.70	89	83	172
2	400±25	0.59	88.5	82	173
3	480±25	0.56	101	91.5	174
4	585±25	0.56	123	108	174
5	735±25	0.54	149	136	174
6	635±25	0.47	112	105	176
7	740±25	0.445	123	122	176
8	710±25	0.43	114	111	177
9	710±25	0.395	105	107	178
10	790±25	0.375	111	118	179
Tape D					
1	485±40	0.365	67	59	199
2	580±40	0.28	62	60	207
Tape E					
1	570±40	0.225	48	42.5	63
2	735±40	0.20	55	47.5	65

Table 2: Detachment from Teflon

Number of layers <u>N</u>	Pull-off force F/w (N/m)	Pull-off angle <u>θ</u> (rad)	From eq.9	<u>G<sub>a</sub></u> (N/m)	
				From eq.10	From eq.11
<u>Tape A</u>					
1	238±10	0.40	36	37	80
2	297±10	0.34	38	39.5	83
3	325±10	0.295	36	38.5	86
4	382±15	0.27	39	43.5	88
5	435±15	0.26	42	48	89
6	435±20	0.25	41	45	90
7	470±20	0.24	42.5	48	91
8	510±20	0.24	46	51	91
9	530±20	0.24	48	52	91
10	555±20	0.23	47.5	52.5	92
<u>Tape B</u>					
1	525±25	0.175	34.5	34	49
2	725±30	0.155	42	41	49
3	850±30	0.14	44.5	44	49
4	1005±30	0.12	45	50	49
5	1080±30	0.12	48.5	51.5	49
6	1145±40	0.115	49.5	52	49
7	1195-40	0.105	47	52.5	49
8	1275±75	0.105	50	54.5	49
9	1275±75	0.105	50	52.5	49
10	1315±75	0.095	47	53	49
<u>Tape C</u>					
1	89±4	0.42	14.0	13.0	24
2	116±8	0.35	15.2	14.7	24
3	124±8	0.305	14.2	14.1	24
4	151±8	0.28	15.9	16.6	24
5	170±8	0.26	16.6	18.1	24
6	182±8	0.245	16.7	18.6	24
7	185±8	0.22	15.3	18.1	24
8	193±8	0.21	15.2	18.3	24
9	208±8	0.19	14.8	19.6	24
10	228±8	0.185	15.8	21.2	24

Figure Captions

- Figure 1. Sketch of the pull-off experiment.
- Figure 2. Experimental relations between tensile load per unit width  $F/w$  and extension  $e$  for selected tapes.  
(a) Tapes A and C  
(b) Tape B
- Figure 3. Peel experiment
- Figure 4. Plot of the pull-off force per unit width  $F/w$  vs  $N^{\frac{1}{4}}$  where  $N$  is the number of layers of tape applied one on top of another to a glass substrate and pulled away together.
- Figure 5. Plot of the pull-off force per unit width  $F/w$  vs  $N^{\frac{1}{4}}$  where  $N$  is the number of layers of tape applied one on top of another to a Teflon substrate and pulled away together.
- Figure 6. Plot of  $F_0/w$  vs  $N$  for Tape A adhering to glass (open circles) and to Teflon (filled-in circles).
- Figure 7. Plot of  $F_0/w$  vs  $N$  for Tape B adhering to glass (open triangles) and to Teflon (filled-in triangles).
- Figure 8. Plot of  $F_0/w$  vs  $N$  for Tape C adhering to glass (open squares) and to Teflon (filled-in squares).

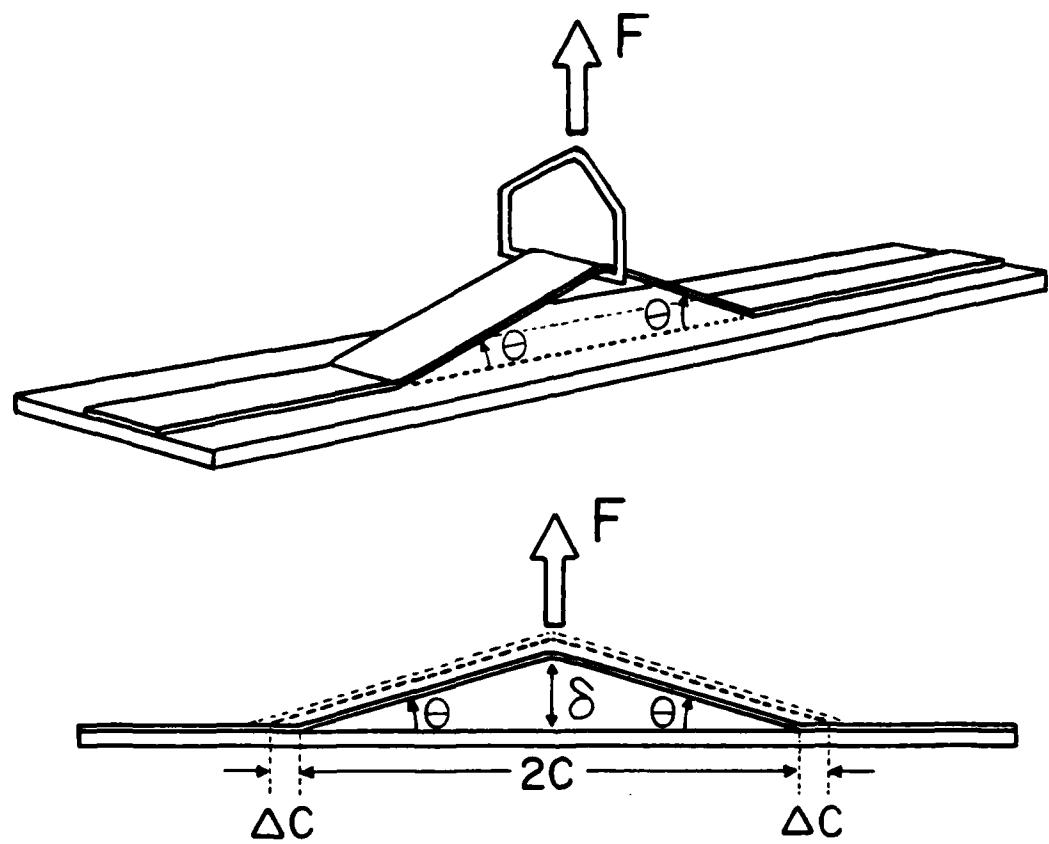


Figure 1

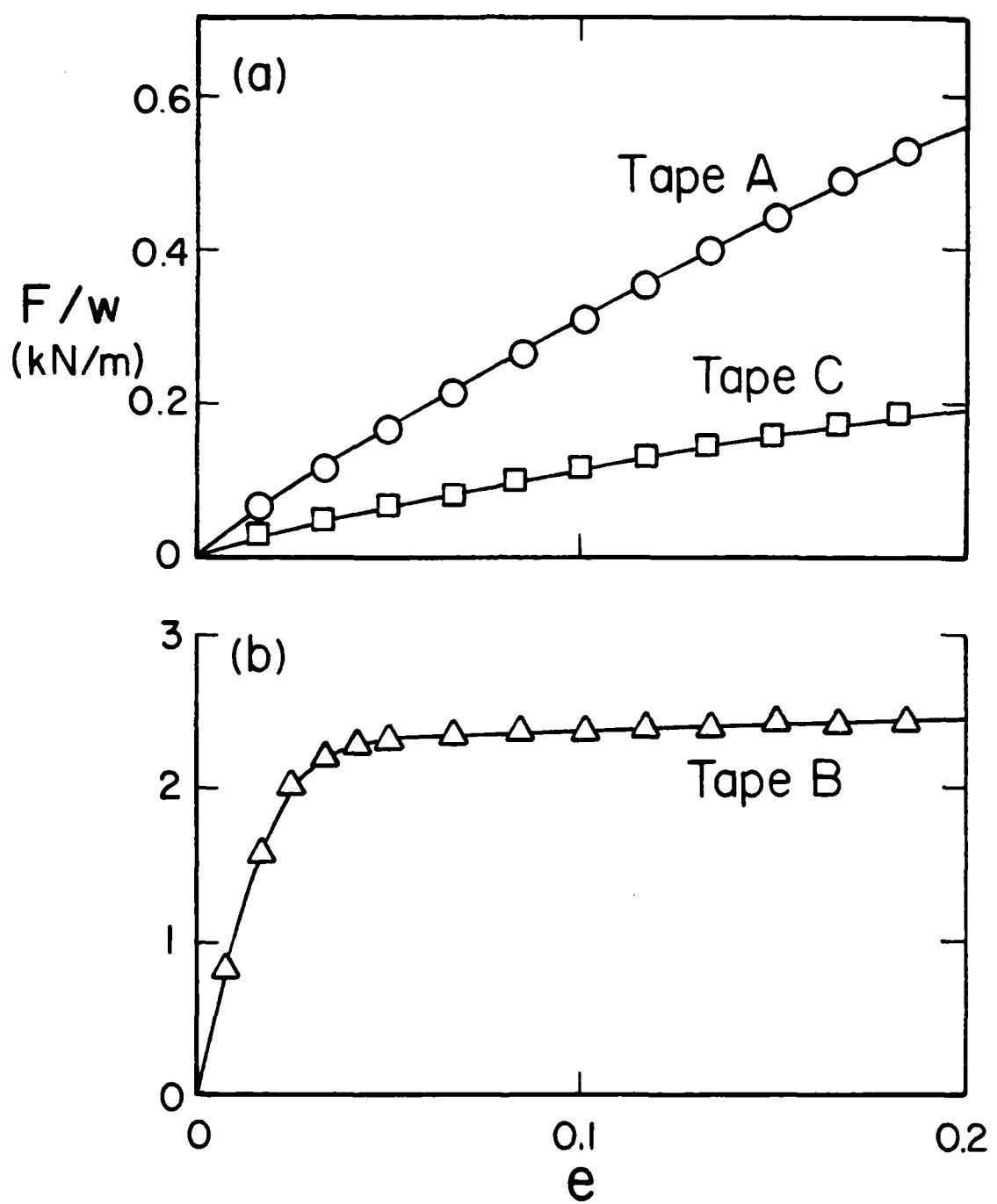


Figure 2

21.

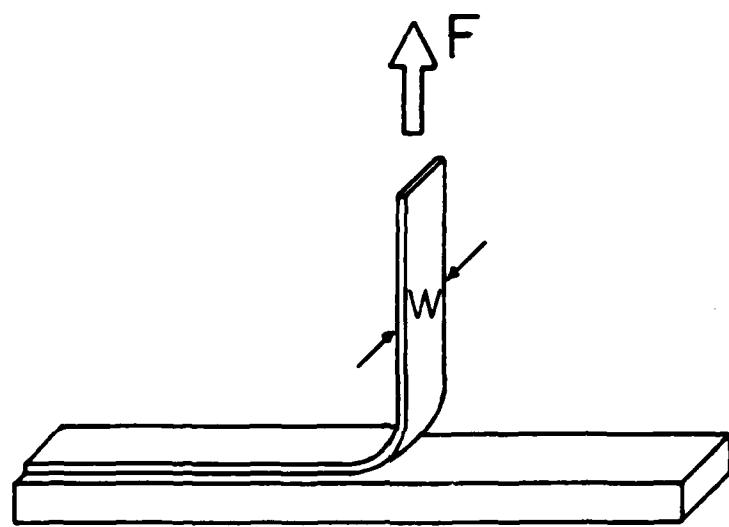


Figure 3

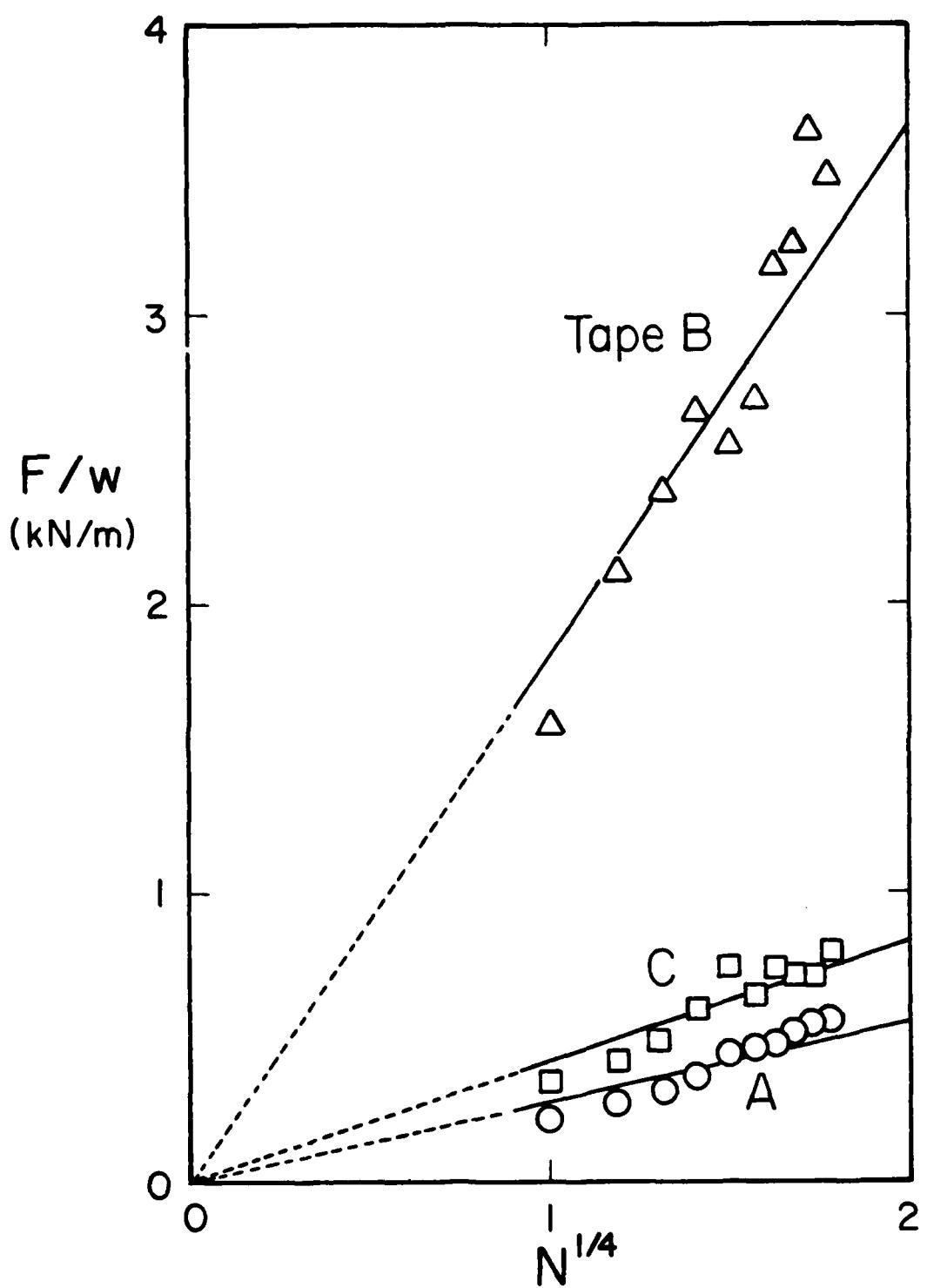


Figure 4

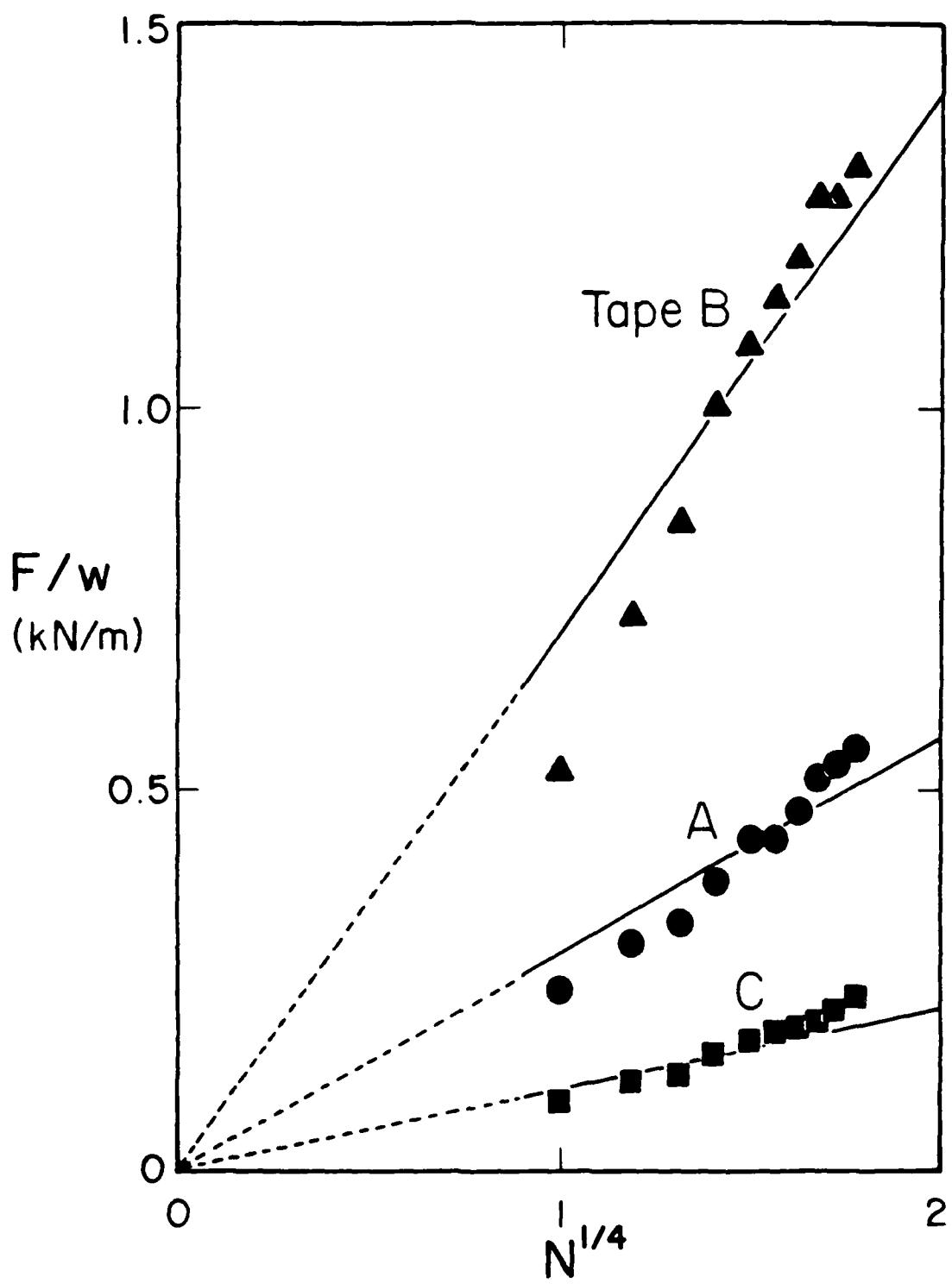


Figure 5

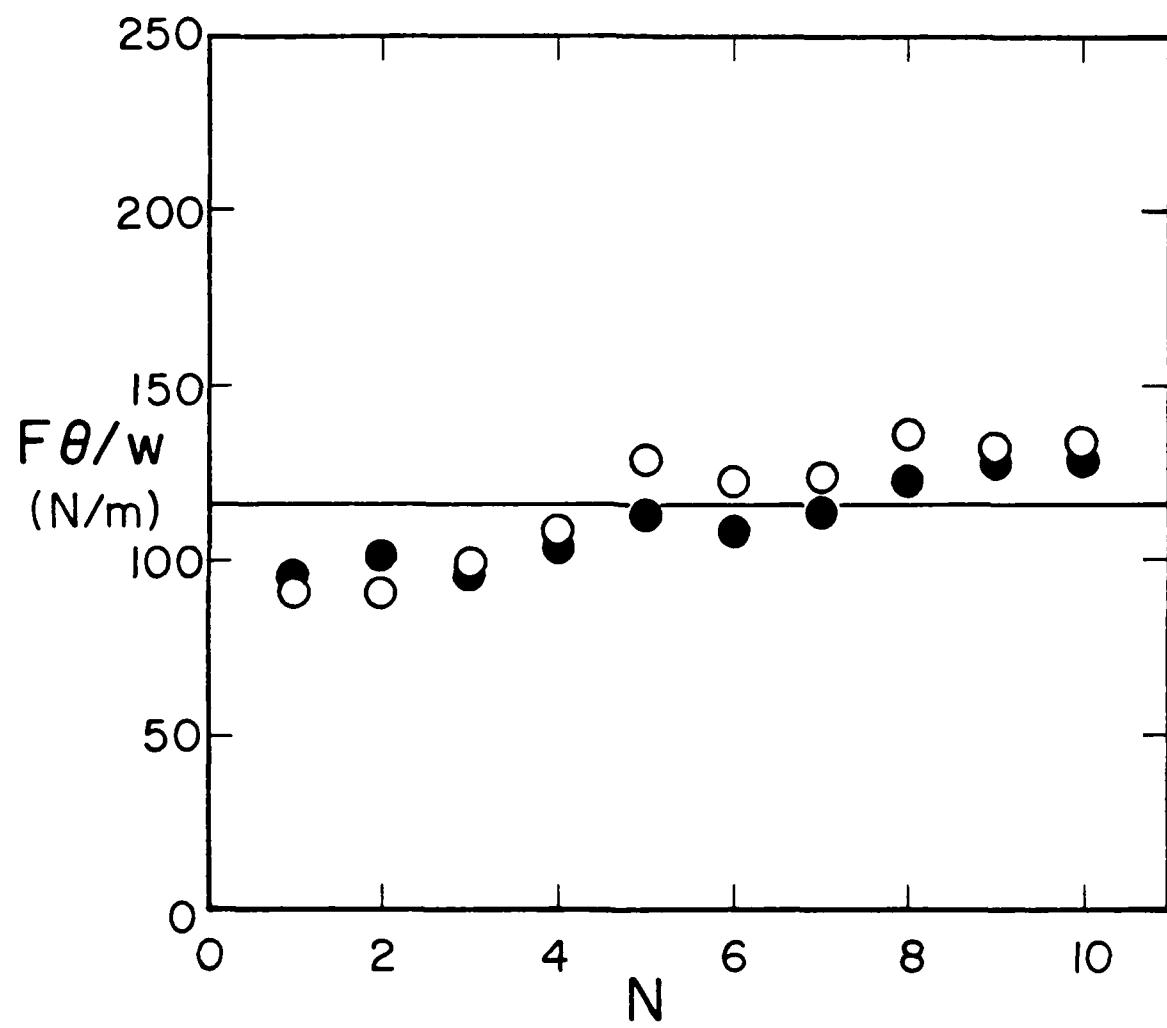


Figure 6

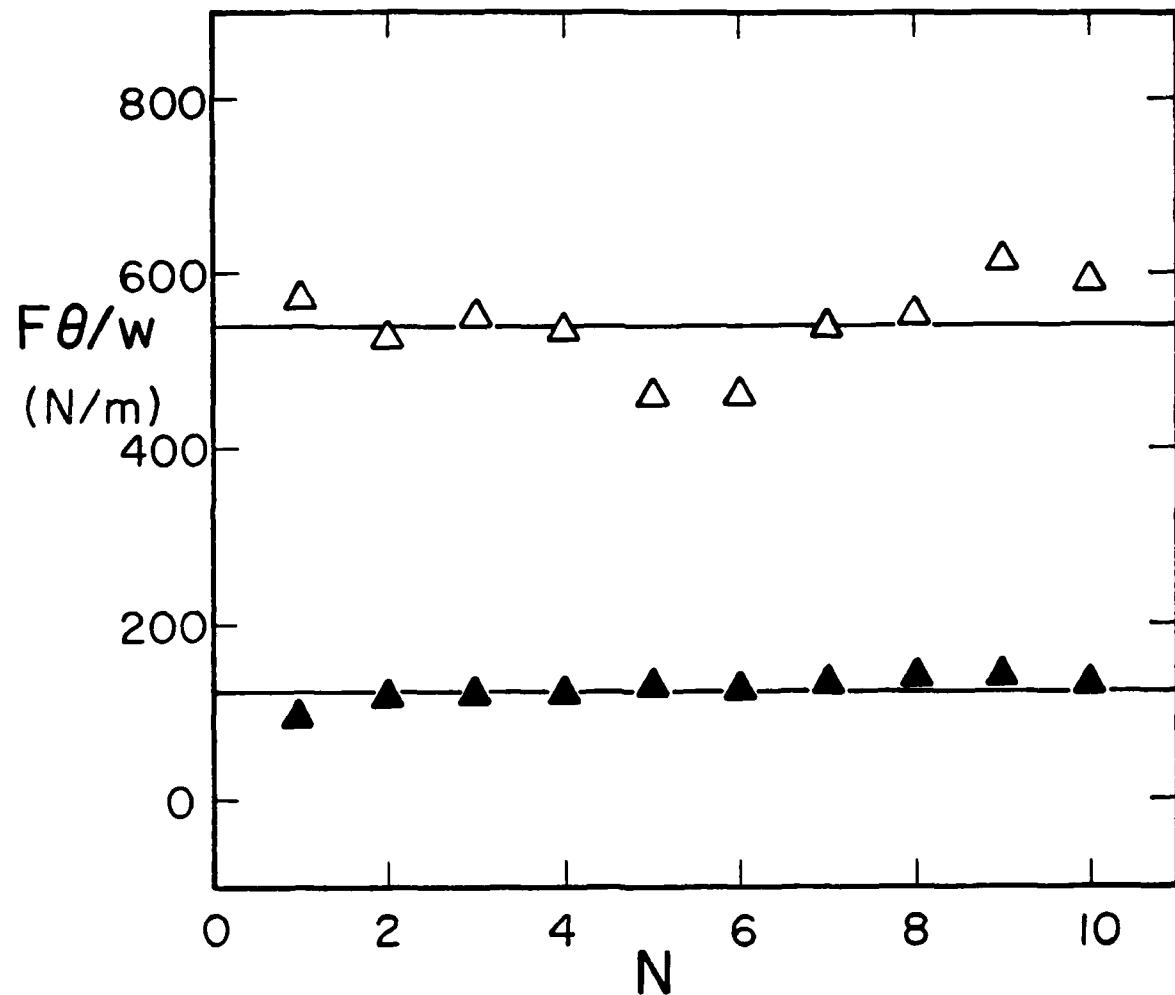


Figure 7

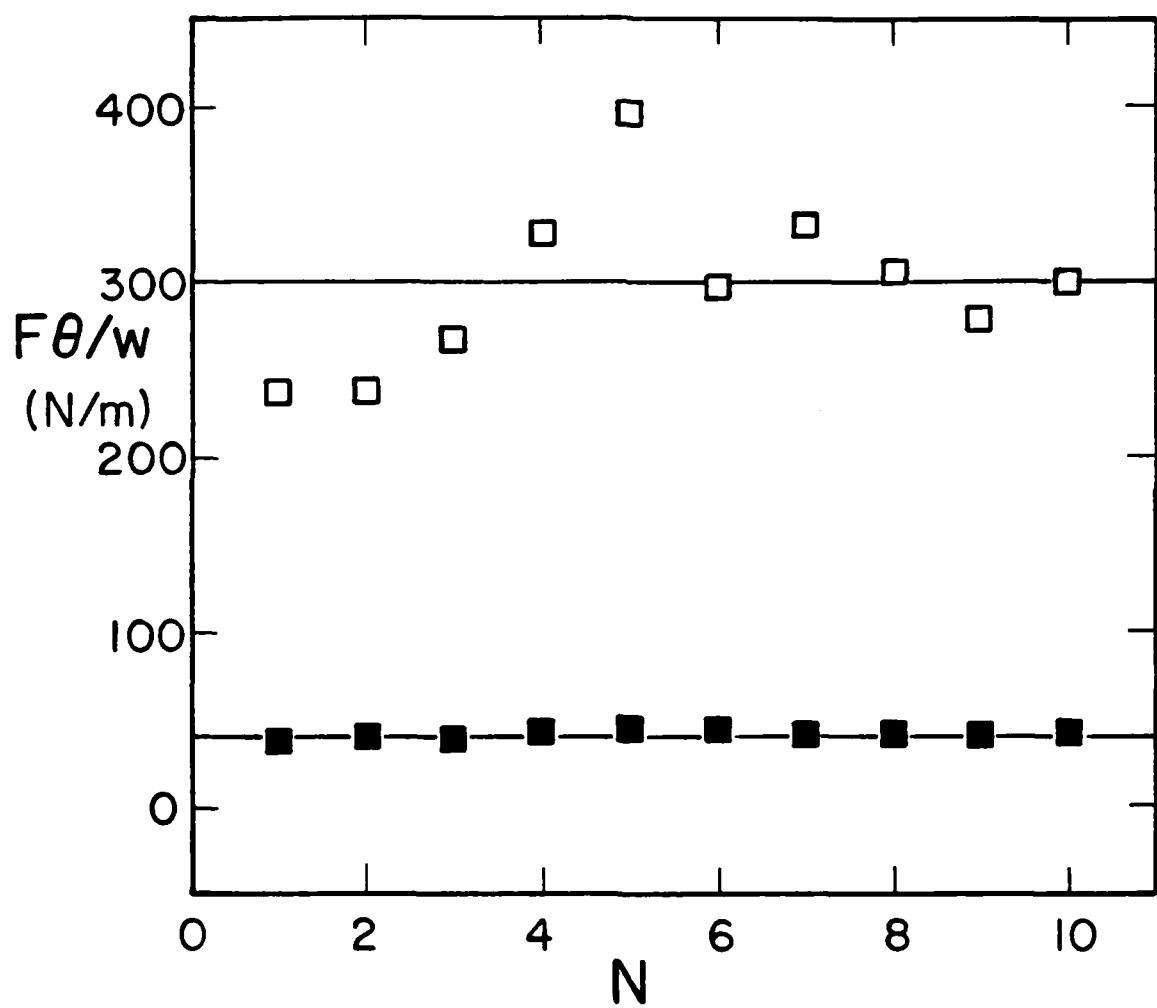


Figure 8

(DYN)

27.

DISTRIBUTION LIST

Dr. R.S. Miller  
Office of Naval Research  
Code 432P  
Arlington, VA 22217  
(10 copies)

Dr. J. Pastine  
Naval Sea Systems Command  
Code 06R  
Washington, DC 20362

Dr. Kenneth D. Hartman  
Hercules Aerospace Division  
Hercules Incorporated  
Alleghany Ballistic Lab  
P.O. Box 210  
Washington, DC 21502

Mr. Otto K. Heiney  
AFATL-DLJG  
Elgin AFB, FL 32542

Dr. Merrill K. King  
Atlantic Research Corp.  
5390 Cherokee Avenue  
Alexandria, VA 22312

Dr. R.L. Lou  
Aerojet Strategic Propulsion Co.  
Bldg. 05025 - Dept 5400 - MS 167  
P.O. Box 15699C  
Sacramento, CA 95813

Dr. R. Olsen  
Aerojet Strategic Propulsion Co.  
Bldg. 05025 - Dept 5400 - MS 167  
P.O. Box 15699C  
Sacramento, CA 95813

Dr. Randy Peters  
Aerojet Strategic Propulsion Co.  
Bldg. 05025 - Dept 5400 - MS 167  
P.O. Box 15699C  
Sacramento, CA 95813

Dr. D. Mann  
U.S. Army Research Office  
Engineering Division  
Box 12211  
Research Triangle Park, NC 27709-2211

Dr. L.V. Schmidt  
Office of Naval Technology  
Code 07CT  
Arlington, VA 22217

JHU Applied Physics Laboratory  
ATTN: CPIA (Mr. T.W. Christian)  
Johns Hopkins Rd.  
Laurel, MD 20707

Dr. R. McGuire  
Lawrence Livermore Laboratory  
University of California  
Code L-324  
Livermore, CA 94550

P.A. Miller  
736 Leavenworth Street, #6  
San Francisco, CA 94109

Dr. W. Moniz  
Naval Research Lab.  
Code 6120  
Washington, DC 20375

Dr. K.F. Mueller  
Naval Surface Weapons Center  
Code R11  
White Oak  
Silver Spring, MD 20910

Prof. M. Nicol  
Dept. of Chemistry & Biochemistry  
University of California  
Los Angeles, CA 90024

Mr. L. Roslund  
Naval Surface Weapons Center  
Code R10C  
White Oak, Silver Spring, MD 20910

Dr. David C. Sayles  
Ballistic Missile Defense  
Advanced Technology Center  
P.O. Box 1500  
Huntsville, AL 35807

(DYN)

DISTRIBUTION LIST

Mr. R. Geisler  
ATTN: DY/MS-24  
AFRPL  
Edwards AFB, CA 93523

Naval Air Systems Command  
ATTN: Mr. Bertram P. Sobers  
NAVAIR-320G  
Jefferson Plaza 1, RM 472  
Washington, DC 20361

R.B. Steele  
Aerojet Strategic Propulsion Co.  
P.O. Box 15699C  
Sacramento, CA 95813

Mr. M. Stosz  
Naval Surface Weapons Center  
Code R10B  
White Oak  
Silver Spring, MD 20910

Mr. E.S. Sutton  
Thiokol Corporation  
Elkton Division  
P.O. Box 241  
Elkton, MD 21921

Dr. Grant Thompson  
Morton Thiokol, Inc.  
Wasatch Division  
MS 240 P.O. Box 524  
Brigham City, UT 84302

Dr. R.S. Valentini  
United Technologies Chemical Systems  
P.O. Box 50015  
San Jose, CA 95150-0015

Dr. R.F. Walker  
Chief, Energetic Materials Division  
DRSMC-LCE (D), B-3022  
USA ARDC  
Dover, NJ 07801

Dr. Janet Wall  
Code 012  
Director, Research Administration  
Naval Postgraduate School  
Monterey, CA 93943

Director  
US Army Ballistic Research Lab.  
ATTN: DRXBR-IBD  
Aberdeen Proving Ground, MD 21005

Commander  
US Army Missile Command  
ATTN: DRSMI-RKL  
Walter W. Wharton  
Redstone Arsenal, AL 35898

Dr. Ingo W. May  
Army Ballistic Research Lab.  
ARRADCOM  
Code DRXBR - 1BD  
Aberdeen Proving Ground, MD 21005

Dr. E. Zimet  
Office of Naval Technology  
Code 071  
Arlington, VA 22217

Dr. Ronald L. Derr  
Naval Weapons Center  
Code 389  
China Lake, CA 93555

T. Boggs  
Naval Weapons Center  
Code 389  
China Lake, CA 93555

Lee C. Estabrook, P.E.  
Morton Thiokol, Inc.  
P.O. Box 30058  
Shreveport, Louisiana 71130

Dr. J.R. West  
Morton Thiokol, Inc.  
P.O. Box 30058  
Shreveport, Louisiana 71130

Dr. D.D. Dillehay  
Morton Thiokol, Inc.  
Longhorn Division  
Marshall, TX 75670

G.T. Bowman  
Atlantic Research Corp.  
7511 Wellington Road  
Gainesville, VA 22065

(DYN)

29.

DISTRIBUTION LIST

R.E. Shenton  
Atlantic Research Corp.  
7511 Wellington Road  
Gainesville, VA 22065

Mike Barnes  
Atlantic Research Corp.  
7511 Wellington Road  
Gainesville, VA 22065

Dr. Lionel Dickinson  
Naval Explosive Ordnance  
Disposal Tech. Center  
Code D  
Indian Head, MD 20340

Prof. J.T. Dickinson  
Washington State University  
Dept. of Physics 4  
Pullman, WA 99164-2814

M.H. Miles  
Dept. of Physics  
Washington State University  
Pullman, WA 99164-2814

Dr. T.F. Davidson  
Vice President, Technical  
Morton Thiokol, Inc.  
Aerospace Group  
110 North Wacker Drive  
Chicago, Illinois 60606

Mr. J. Consaga  
Naval Surface Weapons Center  
Code R-16  
Indian Head, MD 20640

Naval Sea Systems Command  
ATTN: Mr. Charles M. Christensen  
NAVSEA-62R2  
Crystal Plaza, Bldg. 6, Rm 806  
Washington, DC 20362

Mr. R. Beauregard  
Naval Sea Systems Command  
SEA 64E  
Washington, DC 20362

Brian Wheatley  
Atlantic Research Corp.  
7511 Wellington Road  
Gainesville, VA 22065

Mr. G. Edwards  
Naval Sea Systems Command  
Code 62R32  
Washington, DC 20362

C. Dickinson  
Naval Surface Weapons Center  
White Oak, Code R-13  
Silver Spring, MD 20910

Prof. John Deutch  
MIT  
Department of Chemistry  
Cambridge, MA 02139

Dr. E.H. deButts  
Hercules Aerospace Co.  
P.O. Box 27408  
Salt Lake City, UT 84127

David A. Flanigan  
Director, Advanced Technology  
Morton Thiokol, Inc.  
Aerospace Group  
110 North Wacker Drive  
Chicago, Illinois 60606

Dr. L.H. Caveny  
Air Force Office of Scientific  
Research  
Directorate of Aerospace Sciences  
Bolling Air Force Base  
Washington, DC 20332

W.G. Roger  
Code 5253  
Naval Ordnance Station  
Indian Head, MD 20640

Dr. Donald L. Ball  
Air Force Office of Scientific  
Research  
Directorate of Chemical &  
Atmospheric Sciences  
Bolling Air Force Base  
Washington, DC 20332

(DYN)

DISTRIBUTION LIST

Dr. Anthony J. Matuszko  
Air Force Office of Scientific Research  
Directorate of Chemical & Atmospheric  
Sciences  
Bolling Air Force Base  
Washington, DC 20332

Dr. Michael Chaykovsky  
Naval Surface Weapons Center  
Code R11  
White Oak  
Silver Spring, MD 20910

J.J. Rocchio  
USA Ballistic Research Lab.  
Aberdeen Proving Ground, MD 21005-5066

G.A. Zimmerman  
Aerojet Tactical Systems  
P.O. Box 13400  
Sacramento, CA 95813

B. Swanson  
INC-4 MS C-346  
Los Alamos National Laboratory  
Los Alamos, New Mexico 87545

Dr. James T. Bryant  
Naval Weapons Center  
Code 3205B  
China Lake, CA 93555

Dr. L. Rothstein  
Assistant Director  
Naval Explosives Dev. Engineering Dept.  
Naval Weapons Station  
Yorktown, VA 23691

Dr. M.J. Kamlet  
Naval Surface Weapons Center  
Code R11  
White Oak, Silver Spring, MD 20910

Dr. Henry Webster, III  
Manager, Chemical Sciences Branch  
ATTN: Code 5063  
Crane, IN 47522

Dr. A.L. Slafkosky  
Scientific Advisor  
Commandant of the Marine Corps  
Code RD-1  
Washington, DC 20380

Dr. H.G. Adolph  
Naval Surface Weapons Center  
Code R11  
White Oak  
Silver Spring, MD 20910

U.S. Army Research Office  
Chemical & Biological Sciences  
Division  
P.O. Box 12211  
Research Triangle Park, NC 27709

G. Butcher  
Hercules, Inc.  
MS X2H  
P.O. Box 98  
Magna, Utah 84044

W. Waesche  
Atlantic Research Corp.  
7511 Wellington Road  
Gainesville, VA 22065

Dr. John S. Wilkes, Jr.  
FJSRL/NC  
USAF Academy, CO 80840

Dr. H. Rosenwasser  
AIR-320R  
Naval Air Systems Command  
Washington, DC 20361

Dr. Joyce J. Kaufman  
The Johns Hopkins University  
Department of Chemistry  
Baltimore, MD 21218

Dr. A. Nielsen  
Naval Weapons Center  
Code 385  
China Lake, CA 93555

(DYN)

DISTRIBUTION LIST

K.D. Pae  
High Pressure Materials Research Lab.  
Rutgers University  
P.O. Box 909  
Piscataway, NJ 08854

Dr. John K. Dienes  
T-3, B216  
Los Alamos National Lab.  
P.O. Box 1663  
Los Alamos, NM 87544

A.N. Gent  
Institute Polymer Science  
University of Akron  
Akron, OH 44325

Dr. D.A. Shockey  
SRI International  
333 Ravenswood Ave.  
Menlo Park, CA 94025

Dr. R.B. Kruse  
Morton Thiokol, Inc.  
Huntsville Division  
Huntsville, AL 35807-7501

G. Butcher  
Hercules, Inc.  
P.O. Box 98  
Magna, UT 84044

W. Waesche  
Atlantic Research Corp.  
7511 Wellington Road  
Gainesville, VA 22065

Dr. R. Bernecker  
Naval Surface Weapons Center  
Code R13  
White Oak  
Silver Spring, MD 20910

Prof. Edward Price  
Georgia Institute of Tech.  
School of Aerospace Engineering  
Atlanta, GA 30332

J.A. Birkett  
Naval Ordnance Station  
Code 5253K  
Indian Head, MD 20640

Prof. R.W. Armstrong  
University of Maryland  
Dept. of Mechanical Engineering  
College Park, MD 20742

Herb Richter  
Code 385  
Naval Weapons Center  
China Lake, CA 93555

J.T. Rosenberg  
SRI International  
333 Ravenswood Ave.  
Menlo Park, CA 94025

G.A. Zimmerman  
Aeroject Tactical Systems  
P.O. Box 13400  
Sacramento, CA 95813

Prof. Kenneth Kuo  
Pennsylvania State University  
Dept. of Mechanical Engineering  
University Park, PA 16802

T.L. Boggs  
Naval Weapons Center  
Code 3891  
China Lake, CA 93555

(DYN)

32.

DISTRIBUTION LIST

Dr. C.S. Coffey  
Naval Surface Weapons Center  
Code R13  
White Oak  
Silver Spring, MD 20910

D. Curran  
SRI International  
333 Ravenswood Avenue  
Menlo Park, CA 94025

E.L. Throckmorton  
Code SP-2731  
Strategic Systems Program Office  
Crystal Mall #3, RM 1048  
Washington, DC 23076

Dr. R. Martinson  
Lockheed Missiles and Space Co.  
Research and Development  
3251 Hanover Street  
Palo Alto, CA 94304

C. Gotzmer  
Naval Surface Weapons Center  
Code R-11  
White Oak  
Silver Spring, MD 20910

G.A. Lo  
3251 Hanover Street  
B204 Lockheed Palo Alto Research Lab  
Palo Alto, CA 94304

R.A. Schapery  
Civil Engineering Department  
Texas A&M University  
College Station, TX 77843

J.M. Culver  
Strategic Systems Projects Office  
SSPO/SP-2731  
Crystal Mall #3, RM 1048  
Washington, DC 20376

Prof. G.D. Duvall  
Washington State University  
Department of Physics  
Pullman, WA 99163

Dr. E. Martin  
Naval Weapons Center  
Code 3858  
China Lake, CA 93555

Dr. M. Farber  
135 W. Maple Avenue  
Monrovia, CA 91016

W.L. Elban  
Naval Surface Weapons Center  
White Oak, Bldg. 343  
Silver Spring, MD 20910

G.E. Manser  
Morton Thiokol  
Wasatch Division  
P.O. Box 524  
Brigham City, UT 84302

R.G. Rosemeier  
Brimrose Corporation  
7720 Belair Road  
Baltimore, MD 20742

33.

**Administrative Contracting  
Officer (see contract for  
address)  
(1 copy)**

**Director  
Naval Research Laboratory  
Attn: Code 2627  
Washington, DC 20375  
(6 copies)**

**Defense Technical Information Center  
Bldg. 5, Cameron Station  
Alexandria, VA 22314  
(12 copies)**

**Dr. Robert Polvani  
National Bureau of Standards  
Metallurgy Division  
Washington, D.C. 20234**

**Dr. Y. Gupta  
Washington State University  
Department of Physics  
Pullman, WA 99163**

END  
FILED

5-86

DTIC